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DOGUMENT NO & TYPE	SUBJECT/TITLE	THE DATE	RESTRICTION
16. note 17. paper 18. paper 19. charts	JP to David, re: Bud's decision, 2p 11 16 81 F98-054 #81 10 12 6 95 re: Other aircraft, 1p re: Taiwan Strait, 8p 78-054 #83 re: LWDF, 9p 78-054 #83	nd / nd / nd / nd / nd	P1/F1 # 29 P1/F1 # 31 P1/F1 # 31
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RESTRICTIONS

- P-1 National security classified information [(a)(1) of the PRA]. P-2 Relating to appointment to Federal office [(a)(2) of the PRA].
- P-3 Release would violate a Federal statute [(a)(3) of the PRA].
- P-4 Release would disclose trade secrets or confidential commercial or financial information [(a)(4) of the PRA].
- P-5 Release would disclose confidential advice between the President and his advisors, or between such advisors [(a)(5) of the PRA].
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- F-7 Release would disclose information compiled for law enforcement purposes [(b)(7) of the FOIA].
- F-8 Release would disclose information concerning the regulation of financial institutions [(b)(8) of the FOIA].
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DOCUMENTO	SUBJECT/TIFLE	DATE	RESTRICTION
1. report	re: Taiwan 14p S 78 103 198-054 #85 PArt 176	1/15/85	P1/F1 B3
2. memo	R, 6/29/00 NLSF98-054 #86	nd nd	PI/FI
3. paper	re:-executive-summary, 2p	nd -	P1/F1 -
4. paper	re: radar review, 2p P, 6/29/00 NLSF98-054 #88 R 8/24/07 NURL	1143 2	-P1/FI
5. chart	re: radar performance, 1p Part 8/24/07 NURL MO3-1455-# 3	nd	P1/F1
6. chart	R, 6/29/00 NLSF98-054 #90	nd	-P1/F1-
7. chart	re: antenna, 1p R 8/24/07 NLRRM 1455 #4	nd	PIÆ
8. chart	re: aircraft performance, 1p	nd	P1/F1
9-chart	Part 8/24/07 NURR MO3-1455 #5 ro: missions, 1p R. 6/29/00 NUSF98-054 #93	nd	P1/F1
10. paper	re: comments, 1p	nd	P1/F1-
11. draft	Wolfowitz to Armacost re: EAP views [annotated], 5p	nd	P1/F1
12. memo	PATT 12 14 05 MD5-1455 #6 AUD C Wolfowitz/Chain to Armacost re: IDF [annotated], 5p D 5/5/06 NUSF98-054 #96 D NURRY	8/21/84	P1/F1 /107
13. draft memo	Chain/Wolfowitz to Armacost re: assistance [annotated], 4p	nd	P1/F1/ # \$ /!
14/ paper	re: assistance, 2p D 9/5/02 NL 5F98-05H/#98 PATIZITO 105 re: timing, 1p	103-16	55#9
15. paper	re: timing, 1p Dililor 1798-054 #99 PART 1346 105	nd ~_/	#10

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COMPARATIVE AIRCRAFT DATA

Aircrew	LAVI KFIR	F-5E LWD	F F-20A	F-16/79	F-16A	F-16C	F-18	F-15C	M2000C/RDM
Engines/Thrust/(Lbs)	•	2/10,000 (83	50/18,000	1/18,000	1/25,000	1/27,000	2/32,000	2/25,000	1/20,000
Maximum Gross Takeoff Weight (Lbs)	42,000 3570	5 24 ,722	26,950	37,500	37,500	37,500	51,900	68,000	37,787
Fuel Capacity (Lbs)						49.	•		
Internal ·	6,000 5719	4,400	5,050	6,972	6,972	6,972	10,381	13,455	7,275
External	9,180	5,155	6,435	6,760	6,760	6,760	8,280	11,895	2,270
Tanker Compatible	Yes.	Yes	Option	Yes	Yes	Yes	Yes	. Yes	Yes
Thrust/Weight Ratio (50% Internal Fuel)	1.09	.75	1.16	0.81	1.16	1.15	1.06	1.23	1.00

* Garrett, 1042-70

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By MARA, Date 17 6/05

TAIWAN: FIGHTER DEVELOPMENTS (U)

DDB-1300-289-85

Information Cutoff Date: 15 January 1985

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PREFACE

(S/NF/WN/NC) This report discusses the new lightweight defensive fighter being developed by Taiwan to offset the expected decline in its fighter inventory in the next 10 years. This initiative responds to two of Taiwan's perceived problems for the upcoming decade: (1) an increasing threat from the People's Republic of China (PRC) and, (2) increasing difficulty in obtaining foreign, notably US, weapon systems and components. Also discussed are Taiwan's force structuring problem, serospace developments in the PRC affecting Taiwan, and Taiwan's other options for acquiring new fighter aircraft.

The author wishes to especially acknowledge the efforts of Major in compiling and organizing the basic research materials used in this report, and analytical comments and support.

(U) For background concerning PRC tactical air forces, including the potential threat to Taiwan, see China's Tactical Air Forces (U), DDB-1300-281-84, December 1984 and Air Sustainability: Fuzhou Military Region (U), DDB-1300-258-84, August 1984. Also reference China: A Campaign Analysis of Force Projection Capability in the Taiwan Strait (U), Vol I, DDB-2200-29-84, May 1984.

(U) Each classified title and heading in this report have been properly marked; all those unmarked are unclassified.

(U) Word processing support was provided

and

(U) Questions and comments concerning this publication should be referred in writing to the Defense Intelligence Agency (ATTN: DB-2), Washington, D.C. 20301-6111. Requests for additional copies should be forwarded through command approval channels, as appropriate, to DIA (ATTN: RTS-2A), using DD Form 1142, Interagency Document Request, in accordance with DIA Manual 59-3, DIA Reference Library.

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SUMMARY

(S/NF) Taiwan is attempting to design a new lightweight defensive fighter (LDF) aircraft for eventual production. Motivation for this program is the knowledge that the quality of PRC Air Forces will increase significantly in the next decade at the same time that Taiwan's military aircraft inventory will be reduced by the phaseout of obsolescent aircraft. While this program emphasizes indigenous Taiwan production, US Government or corporate advice and technical assistance will still be needed in critical areas such as aircraft engines and avionics.

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1. INTRODUCTION

(S/NF/WN/NC) Taiwan perceives an increasing PRC air threat to its security over the next decade resulting principally from anticipated improvements in the quality of PRC combat aircraft.

As more advanced PRC fighters such as the F-8/FINBACK and improved F-7/FISHBED enter the PRC Inventory, they will reduce the cost of schieving sir superiority if Taiwan's air defenses are not similarly upgraded.

(S/NF/WN/NC) At the same time, Taiwan's options in acquiring advanced foreign equipment (especially US weapons) to balance expected PRC gains have been sharply restricted as a result of political developments. To respond to the expected increase in PRC capabilities and in the face of continuing foreign restrictions, Taiwan has initiated the development of a new lightweight defense fighter.

2. TAIWAN'S NEW FIGHTER

a. Background

(S/NF) To counter the expected increase in PRC capabilities and to offset the degradation of Taipei's aircraft inventory which will surely occur, Taiwan authorities have decided to proceed with the development of a new lightweight defensive fighter (LDF). The diagram shows Taiwan's timeline for the development of the aircraft. Although Taiwan would like to go it alone on the development of the new fighter, this is clearly beyond its capability at this time. A brief discussion of the salient features of the new fighter is given below and highlights those areas where Taiwan will require assistance.

b. New Fighter Attributes

(1) Airframe

-(S/NF/WN) Experience in F-5E/F coproduction and production of its own training aircraft has convinced Taiwan that it could handle LDF airframe development indigenously. Nevertheless, General Dynamics has had a major role in the aircraft's design. Of six original candidates, two airframe configurations are still under consideration: a twin-tail version with widely separated twin engines, and a single-tail version with side-by-side engines. Both versions have a 9.0-meter wingspan and are 13.6 meters long. For comparison, the US F-20 has an 8.1-meter wingspan and is 14.2 meters long. LDF landing gear will be conventional cantilever type, tricycle design.

(2) Engine

-(S/NF/WN/NC) Taiwan has chosen a US corporate associate (Garrett Turbine Engine Corporation) to help in developing the LDF engine, despite its stated intention to maximize self-sufficiency. The selected engine will be the TFE 1042-70, a nonafterburning derivative of the Garrett TFE 731-2. The

4(b) 1.4(d)

latter is a commercial turbojet already used by Taiwan to power its AT-3 trainer. The TFE 1042-70 will be a low-bypass-ratio military engine in the kiloNewton (kN) thrust class.

An afterburning version of the engine in the 53.3 kN thrust range is a likely follow-on. The TFE 1042-70 would be used in the LDF in a twin-engine configuration.

(3)	Flight	Control.	Systems

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-(S/NF/WN/NC) Avionics is the principal remaining area in which evidently still requires considerable foreign assistance and procurement.

To satisfy these criteria, Taiwan probably will contract with a US manufacturer for development of a new radar.

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3. TAIWAN'S PROBLEMS

a. Background

(S/NF) The problems facing Taiwan are two-fold. First, it must find a way of replacing those combat aircraft to be phased out of the inventory in the next 10 years. Second, it must increase the combat capability of its available fighters to better cope with anticipated PRC improvements over the same period.

b. Force Structuring Problem

(S/NF) On paper, Taiwan's current air order of battle picture is the brightest in the past 20 years (table 1). The Air Force's strength will peak by mid-year when coproduction of F-5Es ends and the recently acquired F-104Gs (purchased from the US in 1983) are fully assimilated into the combat force. At that time, the Air Force will have about 400 fighters assigned to combat units with an additional 127 combat-capable fighters assigned to training roles. This growth represents a 57-percent increase in the force since 1976.

1.4 (c)

(S/NF) In the next 10 years, however, the force will decline about as rapidly as it increased. Except for the hewly manufactured P-5E/Fs, the remainder of the fighter inventory is obsolete or rapidly becoming so.

(S/NF) The 57 F-100 A/Fs are all 30 years old and are used only sparingly.

1 it (c)

(S/NF) The 62 F-5A/Bs, despite an excellent maintenance program, are plagued by wing and fuselage cracks and corrosion problems.

1.4(a) 1.4(c)

(S/NF)— In 1983, Taiwan acquired 66 used F-104Gs from the US. These aircraft had been used as trainers by the Federal Republic of Germany Air Force at Luke AFB, Arisona. Taiwan wanted these aircraft to maintain a high performance fighter capability through 1990. The F-104s previously acquired by Taiwan via the Military Assistance Program are all approaching the end of their service life. Even though an in-country Service Life Extension Program (SLEP) was implemented to extend the F-104s' life an additional 1,500 hours, the TAF continues to experience airframe deficiencies, such as cracked I-beams in the tail section. These aircraft probably will be phased out as the recently acquired F-104Gs enter the combat inventory. Even so, these newer aircraft are still 20+ years old and are not expected to have an extended remaining service life of more than 10 years.

c. Attrition Studies

-(3/NF) Since 1982, several studies of TAF attrition have been performed. All of them clearly show a sharp decline in TAF strength over the next 10 years.

An attrition analysis done by OSD/PA&E* in 1982 considered seven different attrition cases. Results ranged from 93 to 185 aircraft lost between 1985 and 1990. The estimates shown in table 1, aircraft by 1990 and by 1995, are based on F-104G readiness rates could be as low as 50 percent by 1995.

4. EXPECTED PRC DEVELOPMENTS

(S/NF) Developments in China's Air Force over the next decade will be geared to meeting the increasingly sophisticated Soviet air threat. It is unlikely that China considers TAF capabilities when programming the design and development of new weapon systems and subsystems. While these new developments may or may not be effective against the Soviet threat, they will increase the CPLAAF's air combat capability relative to Taiwan, if TAF forces decline as projected (see table 2 for comparison of current TAF and PRC air combat aircraft). Among anticipated PRC developments in the next decade, the following are highlighted in table 3 as having the most impact on the TAF:

- -- Significant increase in the number of F-7/FISHBED and F-8/FINBACK interceptors.
- -- Improved air-to-air missiles.
- -- Improved air intercept radars.
- -- Enhanced pilot training programs.

5. TAIWAN'S OTHER OPTIONS

(S/NF) The experience gained in the development of the LDF, even if it is not entirely successful, will be invaluable in establishing a base for the future development of other combat and combat support aircraft. Taiwan officials are pragmatic enough, however, to realize that the LDF is a high-risk venture and that the acquisition of additional fighter aircraft over the next decade is imperative. Table 4 outlines Taiwan's other options for acquiring more fighters.



^{*} Ley Unpublished study done at request of DIA. Study used 1982 order of battle but did account for extended F-5E/F coproduction and acquisition of 66 F-104Gs.

6. CONCLUSION

(S/NF) If Taiwan is successful in producing its own lightweight defensive fighter aircraft, even with engine and avionics assistance from US corporations, it would be a major step forward in armament self-sufficiency for the island. It seems unlikely, however, that initial efforts in this area will be successful quickly enough to eliminate Taiwan's concern-

1.4(3)

Accordingly, further coproduction requests for P-5E/F aircraft or requests to purchase F-20 aircraft or other modern fighters are highly likely.

Table 1
TAF Order of Battle Trend (U)

Combat Fighters	1966	<u>1971</u>	1976	1981	1985	1990 1995 Redacted Redacted
F-86D	14	•	•	-	. •	Do do of o
F-86F	207	79	51	-	-	idRedacted
Ý-5A	· 7	81	48	46	<u>1</u> /	-RedactedRedacted
F-5E	-	-	34	148	248	
F-100A	61	80	79	56	49	ed—Redacted—— 1.4 (b)
F-104A	15	20	17	· 16	15	-Redacted Redacted
F-104D	-	-	2	5	4	Redacted 1.4Cc
F-104G	49	45	29	26	87 <u>3</u> /	ed—Redacted—
			***************************************	***************************************		-Redacted Redacted
•	353	305	260	297	403	edRedacted
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Fighter/Trainers	•	•		1		tedRedacted
F-5A	• -	-	•	-	42	RedactedRedacted
F-5B	2	10	21	20	20	
P-5P		-	-	3	52	for Redacted
F-100F	7	8	10	8	8	Redacted Redacted
F-104B	3	2	. 2	2		acted Ret Record
TF-104G	6	6.	3	3	2 3	To the state of
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	371	331	296	333	530	lacted Rec Redacted
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 $[\]frac{1}{2}$ / Transferred to training role as F-5Es became available.

-s/nf/wn _

^{2/} Production agreement called for F-5Es. Attrition of force is estimated at F-5Es.

^{3/} TAF acquired 66 ex-NATO F-104Gs in 1983.

^{4/} Phaseout of 26 older F-104Gs by 1990 and attrition of remaining aircraft at per year.

^{2/} Production agreement called for F-5F. Attrition estimated at aircraft per year to 1990 and per year to 1995.

Engine (full-scale development began Jun 82) First test flight engines available Jun 86 Engine certified for production Apr 88

Airframe

Conceptual design complete Aug 84

Detailed design & manufacture

Early 85

Prototype rollout Mar 87

Serial production Jan 90

Avionics presently expected to be purchased from US manufacturer rather than indigenously produced

an 84 Jan 85 Jan 86 Jan 87 Jan 88 Jan 89 Jan 90

Figure (S/NR) Timeline for development of lightweight defensive fighter.

Air Combat Comparison (U)

	Air Combat Attribute	TAF PRC TAF PRC TAF F-5E vs F-6 F-5E vs F-7 F-5E vs	The state of the s
	Speed	Redacted Redacted Redacted Redacted	Redacted Red
	Maneuverability	RedactedRedacted	Redacted Red
1	Guns	Redacted Red	Redacted Red
SEC	Missiles*	Redacted Red	Redacted Red
ECRET	Aircraft Radar	Redacted Red	Redacted Redacted Redacted Redacted Redacted
	Pilot Proficiency	Redacted Red	Redacted facted Redacted Redac
	X - Superior in	Redacted Red	edacted Redacted Redacted
		ly equal in this attribute	
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1.4	(a) (c) (d)		

Table 3 PRC Air Combat Improvements (U)

Expected PRC Improvements	Likely Developme	ents	Affect on TAP
- Increase in F-7/FISHBEDs and F-8/FINBACK	<u>1985</u> <u>1990</u>	1995	Redacted Redacted Redacted
- Introduction of F-8-2 in	F-7 240 585 F-8 25 165		Redacted——Redacted——Redacted——Re Redacted——Redacted—
1989 with more advanced radar	F-8-2 0 40 790	300	Redacted—Red
	2/3 /90	1,413	RedactedRedacted
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Improved air-to-air missiles	- Longer-range infrared by 1990 with improved a		Redacted Red
	and more maneuverable t		Redacted——Redacted——Red
· ·	- Radar-guided missile	expected	RedactedRedactedRedactedRedacted
	in mid-1990s.		Redacted—Red
	1	•	Redacted——Redacted——Red
Improved air intercept radars	Longer search/track rSearch/track capabili		RedactedRedactedRedactedRedacted
	- Current F-7:	(Mar)	Redacted Redacted Redacted Red
14 (1)	- Current F-8: - Estimated F-8-2:		RedactedRedactedRedacted
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Inhanced pilot training program	- More information on f capabilities.	•	Redacted Redacted Redacted Red
•	- Increase in dissimila combat training.	r air	Redacted Redacted Redacted Redacted
	- Less reliance on GCI controllers.	ground	-S/NF/VN

Taiwan's Other Options (U)

	Option	Taiwan Preferred	Aircraft Availability	Remarks
	Contined F-5E/F coproduction	No	Negotiable	 Taiwan has stated it does not wish to further coproduce this aircraft Taiwan seeks for current force Improved avionics AIM-9L all-aspect attack missile
o m	Acquire used P-104Gs	No	400-500 still in European inventories	 Interim solution only to longer range problem. Sufficient numbers around to keep TAF inventory near 100 to 1995. All aircraft are 20+ years old, maintenance a problem.
	Acquire advanced US fighters	Yes	Dependent on US political decisi	- Taiwan would prefer F-16/79 but would find F-20 acceptable. - Either sircraft, in quantity, would meet PRC threat for next 20 years. - Favorable US decision could have negative impact on US-PRC relations.
*	Acquire advanced fighters from other Western sources	No.	No direct sales likely. Possibl availability	- TAF has expressed interest in:
			through 3d parti	- Mirage (France) - Tornado (Germany, United Kingdom, Italy) - Piranha (Switzerland) - Kfir, Lavi (Israel) - PRC relations with producers could preclude direct sales

S/NF/WN_

DEUTET



DEPARTMENT OF THE AIR FORCE

OFFICE OF THE ABBITANT SECRETARY

MEMORANDUM FOR ASSISTANT SECRETARY OF DEFENSE (INTERNATIONAL SECURITY AFFAIRS)

SUBJECT: Taiwan's Lightweight Defensive Fighter (LWDF) (Your Memo, Jan 7, 1985) (U) - INFORMATION MEMORANDUM

(8) The results of our initial assessment of Taiwan's LWDF proposal are enclosed for your review (Atch 1-2). Please note that the short suspense precluded a USAF review of the entire LWDF requirements list. Instead, a quick comparison of the Golden Dragon-53 Westinghouse radar and LWDF concept relative to USAF fighter configurations was accomplished. A complete operational assessment could be completed within four months. The comparison draws heavily from US contractor-furnished projected performance data on the engines and radar, since this data was not provided by Taiwan. As such, the Air Force cannot validate this data or the conclusions drawn from them.

(SY The USAF strongly supports the envisaged concept of Taiwan's indigenous fighter aircraft development through US industry license technical assistance agreements. However, continued USAF support to this initiative is contingent upon receipt of the following:

- (U) Formal OSD program direction outlining specific USAF responsibilities and the established operational limitations upon which the aircraft configuration baseline is to be set.
- (%) A Foreign Military Sales administrative case which provides Taiwan funds for USAF manpower, travel and management services attendant with the LWDF initiative.
- Unless otherwise prescribed, we envision the USAF's role in the LWDF program as one of-oversight and interface with the US prime integration contractor of Taiwan's selection. The Office of the Special Assistant for International Cooperative R&D, HQ USAF/RD-I will serve as the Air Staff focal point and facilitate direction and guidance to subordinate commands needed to properly execute the program. Please advise if we can be of further assistance in this matter.

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1. Executive Summary (U)

2. Supporting Documents (8)

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EXECUTIVE SUMMARY (U)

- 1. (S/RF) TASKING: Conduct a quick look comparison of the Westinghouse GD-53 radar versus the APG-63, -66, and -67 radars and LWDF aircraft performance versus F-16A and F-20.
- 2. (U) DATA SOURCES: To conduct this study, data was obtained from the following sources:
- a. (U) Partial engine performance data was obtained from Garrett by telephone request.
- b. (U) F-5, F-16A, F-20 and F-16/J79 performance was generated using existing in-house data.
- c. (U) Aircraft weights were estimated using configuration data from a variety of classified briefing charts, messages, etc.
- d. (U) Aerodynamic data was generated for a generic aircraft representing an F-16 type state-of-the-art fighter.
- e. (U) GD-53 radar performance was estimated based on Westinghouse Proposal (Volume I) for GD-53 Radar System, dated March 1984.
- f. (U) APG-63, -66, and -67 radar performance data was available inhouse.
- 3. (U) AIRCRAFT CONFIGURATION: The LWDF aircraft evaluated was defined as follows:
 - a. (U) Configuration as described in classified briefing charts.
- b. (U) Two (2) Garrett TFE 1042-70 engines each rated at 5000 lb_thrust, 8350 lb augmented.
 - c. (U) Estimated aircraft gross weights (1bs)
 Empty 12,860
 Basic Mission T.O. weight 19,830
 Max 27,200
 - d. (U) Armament 2 AIM-7s, 2 AIM-9s, 7 MK-82s, M61 Gun (500 rounds).
- 4. (U) MISSION PROFILES: The mission profiles used were extracted from in-country briefing material.
- 5. (U) FINDINGS:

a. (e) RADAR - The performance of the GD-53 radar is equivalent to the F-20's APG-67 and is superior to the F-5's APG-159 which does not have a

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look-down capability. With an increase in antenna size the GD-53 could provide performance similar to the F-16's APG-66. From a technical standpoint, the proposed GD-53 development program is considered low risk.

b. (2) PERFORMANCE - For the mission profiles provided, the performance of the aircraft falls approximately midway between the F-5 and the F-20/F-16.

TIDLIVIIAL GD-53 RADAR REVIEW (U)

1. (W) DESCRIPTION:

a. At The GD-53 Radar System as defined in the Westinghouse Defense Electronics Center proposal has been reviewed. The GD-53 radar is an outgrowth or derivative of the USAF AN/APG-66 radar utilized in the F-16A aircraft. The GD-53 utilizes a low sidelobe, two channel monopulse, planar array antenna with IFF dipoles. The antenna is 12.36" high by 19.82" wide; and has a gain of 28.5 db. The GD-53 transmitter is a 21.5 KW peak power traveling wave tube (TWT), air cooled transmitter. The transmitter produces higher power than the APG-66 transmitter. The low power radio frequency (LPRF) unit in the GD-53 is similar to the APG-66, but adds a second receiver channel for two channel monopulse operation. The signal and data processing in the GD-53 is provided by the Radar Digital Assembly (RDA). The RDA uses advanced digital technology, such as configurable gate arrays, large scale integration devices, and flat pack packaging. The RDA uses MIL-STD-1750A Instruction Set Architecture Computers and is programmed in MIL-STD-1589 Jovial Higher Order Language.

- b. (4) The modes available in the GD-53 are shown in Figure 1 along with the modes of other US radars.
- 2. (C) FINDINGS: The proposed radar is an X-band low/medium Pulse Repetition = Frequency pulse doppler radar, which will provide good look-down detection of airborne targets in the presence of ground clutter. The look-down detection performance of this radar is depicted in Figure 2 and is compared to the detection capability of other US radars.: The GD-53 performance is a longless than the APG-66 radar because of a smaller antenna. With equal antenna

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sizes, the GD-53 would provide greater detection range than the APG-66.

The comparisons of antenna gains and average output powers are shown in Figure 3.

3. (C) CONCLUSIONS: The GD-53 is a state-of-the-art pulse doppler radar utilizing the latest electronic devices and digital design. The GD-53 has improved air-to-ground modes over the APG-66 radar; such as, a higher doppler beam sharpening ratio (16:1 vs 8:1) and Ground Moving Target Indication (GMTI). The radar provides added capabilities over the F-5/APG-159 radar; such as look-down air-to-air detection capability and provides approximately 40% more look-up detection range than the APG-159. As shown in Figure 1, the GD-53 provides many more air-to-ground modes than the APG-159. The GD-53 is essentially equivalent to the AN/APQ-67 radar which was developed for the F-20 aircraft.



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RADAR PERFORMANCE COMPARISON

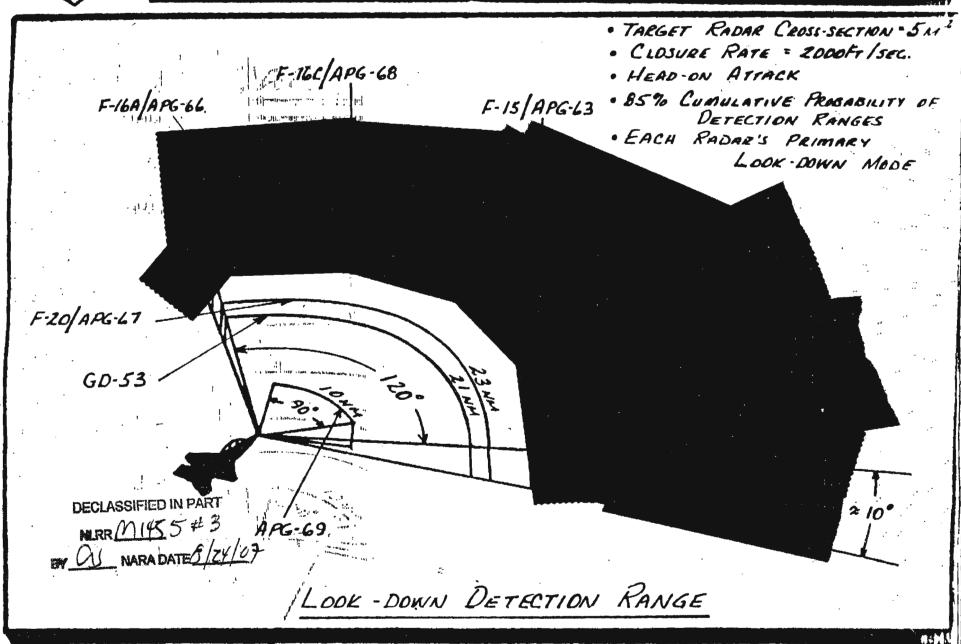


FIGURE 1 AIR-TO-AIR

MODE COMPARISON

AIR-TO-AIR	UNCLASSIFIED WAR PORN DISSE						SEM
!	SEARCH			SINGLE	TRACK	AIR	_
	Look-urin	Look - Down	VELOCITY	TRACK	WHILE	Compat Maneovecous	IFF
GD-53		/	1 , , , , ,	1. 1	~	~	/
APG-63 (F-15)	1. V.	10 10 1 V	' "	V	V	~	V
APG-66 (F-16A)	N. Sept. W.	20 V		~	V	~	
APG-68 (F-16C)		· ·	~	V	V	V	
APG-61 (F-20)	· · · · · · · · · · · · · · · · · · ·	an- char	~	~	~	V.	
APG-69	'	· · · · ·		~		~	
APG-159 (F-5)	V	a.					

IR-TO-SURFACE	REAL BEAM GNO MAP	DOPPLER BEAM SHARPENED MAP	GMTİ	SEARCH	AIR-ID- GROUND RANGING	BEAGON
GD-53	/	16:1	~	/	~	~
APG-63 (F-15)	grade to garage and the control of t	***** ¹ 1			V	/
APG-66 (.F-16A)	1961	E/ 8.1		V	V	V
APG-68 (F-16C)	a stand for an included authorite	64:1	V ;	V	V	~
APG-67 (F-20)*	1 a mar 1/a 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	40:1	V	V	•	
APG-69	r or flowed fire the Charleson had a serv	٠,		·V	V	NO FOR
ARG-159'(F.5)	<i>V</i>		·		·	MA NO.

ANTENNA POWER CHARACTERISTICS

ANTENNA :		DIMENSIONS	1112	GAIN
The state of the s	GD-53	12.36 × 19.82"	245	28.566
• A	PG-63 (F-15)	36" DIAMETER	1018	36.8db
1111 -A	PG-66 (F-KA)	19" x 29"	551	32.6db
,	PG-68 (F-16c)	19" X 29"	551	32.4db
	PG-67 (F-20)	12.5° × 19.8"	548 (320)	29.8db
	PG-69	11.5" X 19"	219	29.5.db
• A	PG-159 (F-5)	11.5" x 19"	219	29.5db

POWER (AVERAGE) =

	• GD -53	238.6 WATTS
	• APG-63 (F-15)	1600 WATTS
	• APG-66 (F-16A)	189.6 WATTS
P	. APG-68 (F-16C)	382 WATTS
ι.	· APG-67 (F-20)	200 WATTS
	· APG-69	120 WATTS
SIFIED STU	· APG-159 (F-5)	60 WATTS
L - 4' 11'.		

BY CAL NARA DATE 8/24/07 LOOK - DOWN MODE FIGURE S

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AIRCRAFT PERFURMANCE ESTIMATES (U)

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10 1		PERFORMANCE PARAMETER	UNITS	LWDF	F-5E	* F-16/ J79		* F-20	
·	MISSION	A. RADIUS OF ACTION	ММ	70	165	140		190.	
	STRIP4''* LAUNCHED INTERCEPT	B. TIME TO CLIMB TO 30,000 FT, MAX POWER	SEC/ MACH	110/.9	90/ .86	75/ .88		50/. 86	
-	2 AIM-7 2 AIM-9 M61 WITH 500 RNDS	C. ACCEL TIME, 0.9M to 1.2M 30,000 FT, 50% INTERNAL FUEL, MAX POWER	SEC	40	75	55		30	
	ra egy ú r targyt e .	D. MAX SUSTAINED G'S M=1.2/30,000 FT, 50% INTERNAL FUEL, MAX POWER	G	3.2	2.1	2.7		3.6	
AIR.		E. MAX MACH/50% INTERNAL FUEL	М	1.9	1.5	1.7		1.9	
AIR-TO-AIR	[!] ને (હા.નો)-	F. Ps 30,000 FT, MAX POWER, 50% INTERNAL FUEL, M=.8/1.4	FT/SEC	240/ 210	180/ 30	190/ 210		310/33 0	
	MISSION II	LOITER TIME, FOR 70NM RADIUS OF ACTION	MIN	100	75	115		105	
	COMBAT AIR PATROL 2 AIM-7 2 AIM-9	音は、3000000			·				
2	2 EXT TANKS M61/500 RNDS	CONTINUE TO A CONTINUE OF ACTION							
< □ 7 MK-82	III	HI-LO-LO-HI RADIUS OF ACTION	NM	88	38	120		110	
	2 AIM-9 M61/500 RNDS	T 1 1 1 1 1 1 1 1 1 1				, , , , , , , , , , , , , , , , , , ,	¢°.		
		+2 5TN 0-10-1	<u> </u>	-	-				

*2 AIM-9s Only See comments on Performance Estimates

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MISSIONS STANDARD DAY ATMOSPHERE

I BASIC

1111 1 1 1 1 1 1 CAP

ARMAMENT: M61A1 20mm gun w/500 rds '

ARMAMENT: M61A1 20mm gun w/500 rds

M61A1 20mm que w/500 rds

2 AIM-7 2 AIM-9 2 AIH-7

2 AIM-9

FUEL: Internal

FUEL: , The entry page forms of 2 External fuel Tanks

Internal

TAKEOFF: Max Power

TAKEOFF: Max Power

CL IMB:

.1 .

Max Power

15 H, CLIMB:

Mil Power

CRUISE

Out at 36,000 ft ALTITUDE: In at optimum

CRUISE

Opt imum

ALTITUDE: COMBAT:

Opt imum

COMBAT:

FUEL

15,000 ft/M=0.8

the contract of a contract of the contract of

15,000 ft/M=0,B

5 minutes at Max Power. Fire Missiles

5 minutes at Max Power.

Fire Missiles

10 minutes endurance at

PMR

COMBAT

10 minutes endurance at

RESERVES: sea level

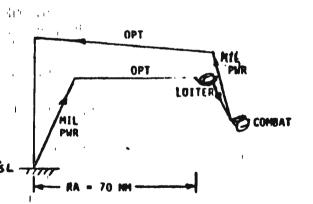
11 11 1

OPT

... 36,000 FT

'FUEL

RESERVES: sea'level



III AIR-TO-GROUND

ARMAMENT:

2 AIM-9

7 MK-82 Bombs

FUEL: Internal

2 External fuel Tanks

TAKEOFF: Max Power

CL IMB: 1 Max Power

CRUISE 1

Opt imum

ALTITUDE: Opt Imum

COMBAT: Sea Level/M-0.9

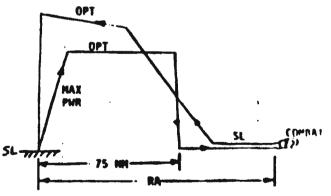
5 minutes at Max Power.

Drop Bombs

FUEL

10 minutes endurance at

RESERVES: sea level



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COMMENTS ON PERFORMANCE ESTIMATES

- 1. Mission profiles developed by the Air Force are normally task oriented to maximize performance with the least amount of fuel. The LWDF Mission Profiles are unique in that they require max use of afterburner for take-off and climb due to the short radius of action requirement and five minutes of afterburner for combat at midpoint. Consequently, the use of these same profiles for the comparative aircraft results in a large expenditure of fuel due to their bigger engines (e.g., for the F-16A, over 50% of the internal fuel is used for the 5 minutes combat time in Mission I). The result is that the performance numbers shown are significantly reduced from the capabilities normally associated with these aircraft.
- 2. The conditions specified for the performance parameters were selected on the basis of the estimated engine data handcarried to ASD by Garrett. Due to the time constraint, no attempt was made to determine the validity of this data.
- 3. For the Radius of Action parameter, only two AIM-9s were used for the comparative aircraft, since they are not currently authorized the carriage of AIM-7s.
- 4. Ps (Excess Specific Power) is defined as a value representing the aircraft's ability to accelerate and/or climb at a specific point in time along the velocity axis.
- 5. Comparative aircraft data used were representative of current production aircraft configuration and existing flight manual limitations. No attempt was made to optimize the comparative aircraft configuration for the three basic missions.
- 6. Estimates for the LWDF were based on the in-country briefing data package for weights, wing area and configuration and aerodynamic drag trends based on F-16 drag data.

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